DEVELOPING A MATHEMATICAL MODEL OF AIRCRAFT HONEYCOMB SANDWICHES

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A vibrational model of a three- layer honeycomb sandwich with partial skin peeling has been developed. The impact of peeling amount on the natural frequency of vibration and the amplitude-frequency characteristic has been analysed. Comparative analysis of design values and experimentally obtained results has been made.

Honeycomb sandwiches, multilayer panels, peeling, vibration theory, technical condition control.

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